

IEE International Frequency Control Symposium 1998

Commercial Off-The-Shelf (COTS)

**A Study of Plastic Encapsulated
Microcircuits (PEMs) in JPL Space Hardware**

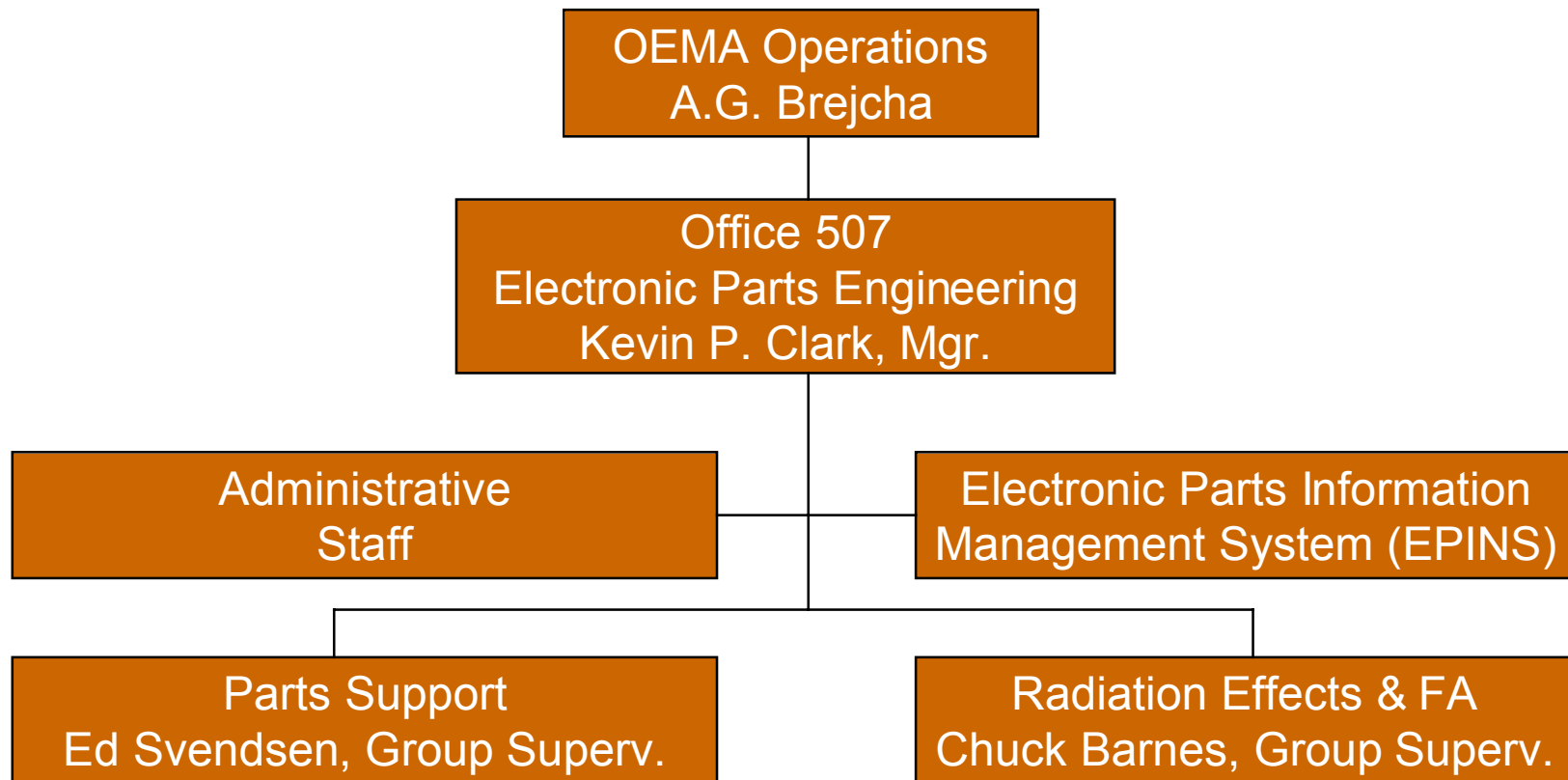


Mike Sandor & Shri Agarwal

JET PROPULSION LABORATORY
Electronic Parts Engineering Office



Organization



A decorative horizontal bar with a gradient from dark blue to yellow, ending in a pointed, comet-like shape on the right side.

Agenda

Introduction to COTS

COTS Work Plan/Status

COTS Work for Plastic Packages

Summary



The Meaning of COTS

- “Buy and Fly”
- “Procuring via catalog part number to QML-V standards”
- “Procurement is performed without formal specification”
- “The usage of any COTS equipment does not constitute any waiver to fundamental applicable requirements”

Our Interpretation:

COTS are parts whose specification is manufacturer-controlled as opposed to traditional “Hi-Rel” parts whose specification was Government or customer-controlled



Why Put COTS in Space ?

- 1. The availability of COTS parts is proliferating.**
- 2. COTS parts performance capabilities continue to increase (e.g. processing power & high density memories)**
- 3. A new generation of leading COTS IC technologies is introduced every 3 years.**
- 4. COTS parts typically cost much less than radiation hardened counterparts; by using radiation tolerant parts the cost advantage can be preserved.**
- 5. Some COTS parts (plastic) have been reported to demonstrate good to excellent reliability.**

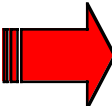
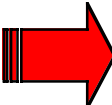
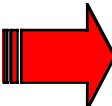
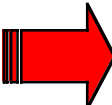
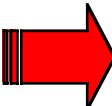
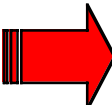


JPL's Concerns About Using COTS

- Reliability/RH of PEMS vs Traditional Ceramic
- Non Rad Hard Designed (maybe Rad Tolerant)
- Narrow Temperature Range
- Process/Designs Change Frequently



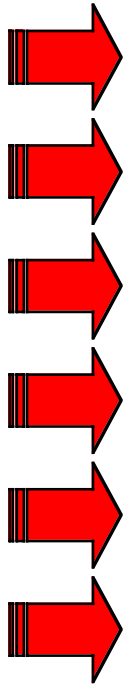
**Examples of Risk Indicators & Their
Relative Costs for a Plastic Package:**

• Temperature Humidity		Corrosion	(\$)
• Temperature Cycling		Assembly Defects	(\$)
• Moisture Absorption		Popcorning	(\$\$)
• Radiation		TID Degradation	(\$\$\$\$)
• Outgassing		Condensables	(\$\$)
• Glass Transition		Stability	(\$\$)

JET PROPULSION LABORATORY
Electronic Parts Engineering Office

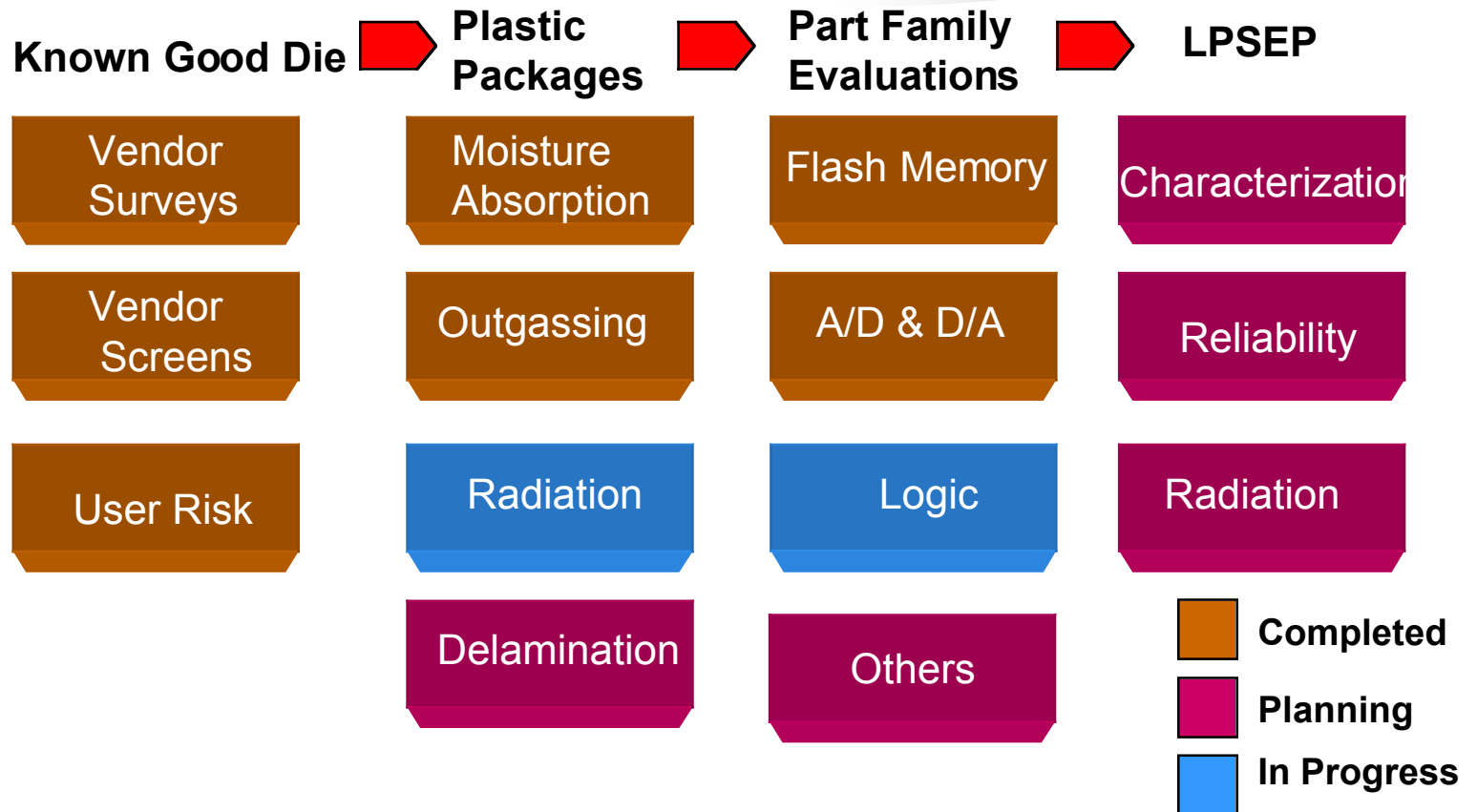


Work Conducted at JPL:

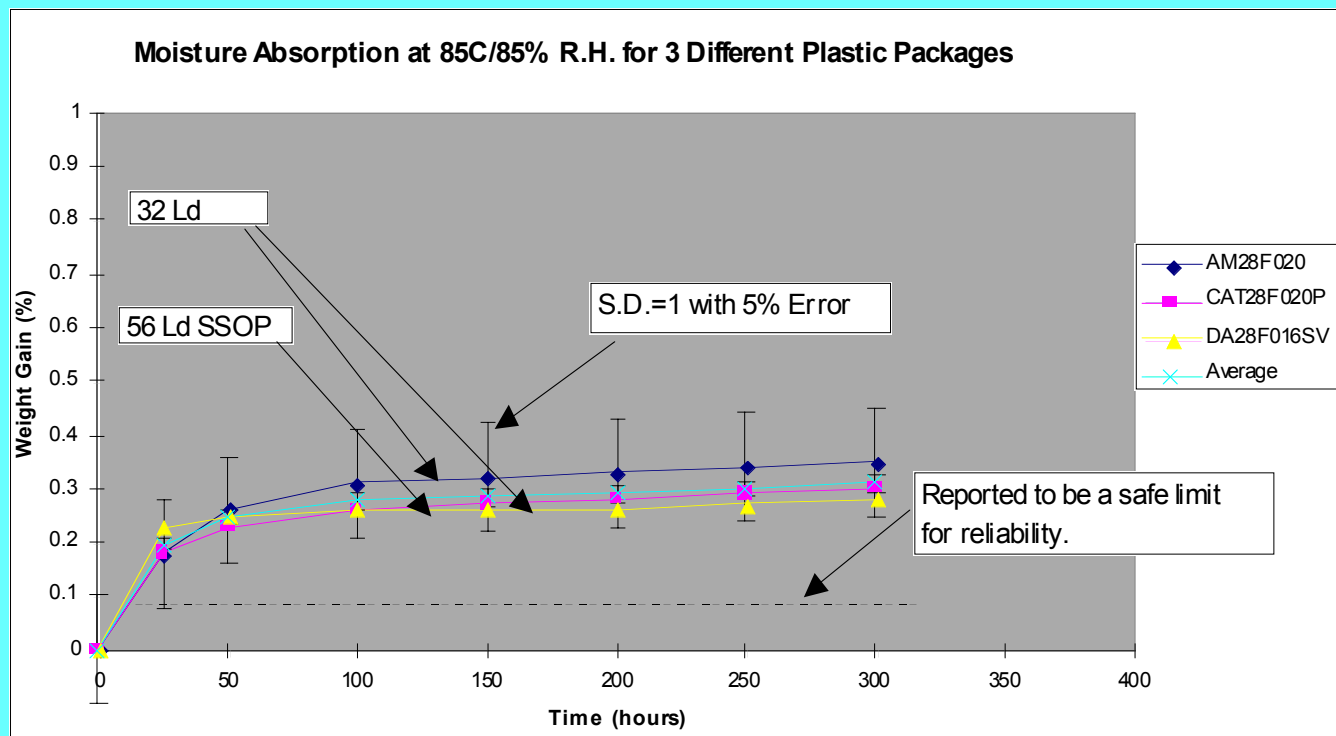




COTS Work- Plan/Status



JET PROPULSION LABORATORY Electronic Parts Engineering Office

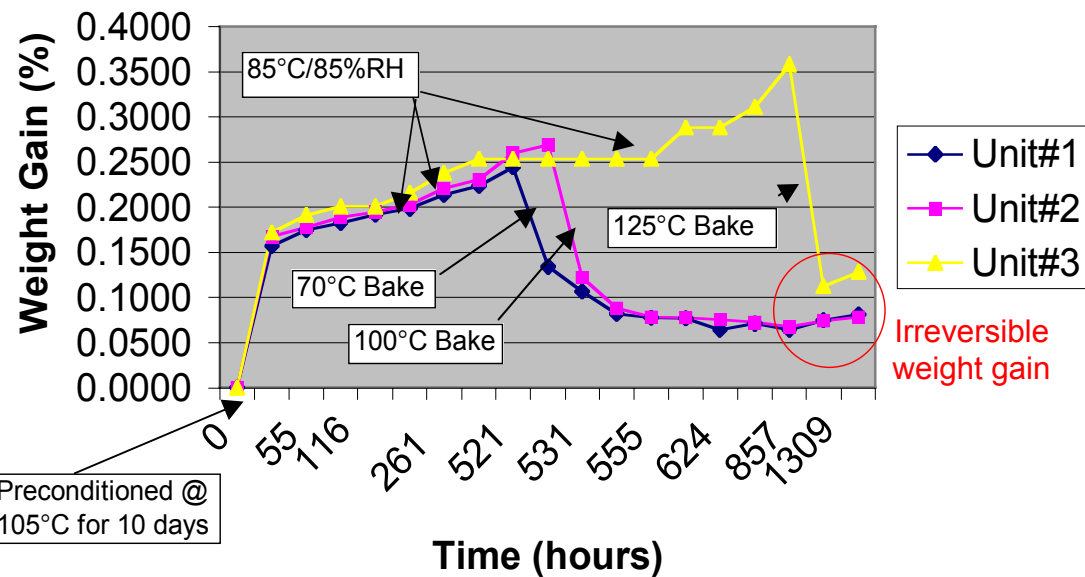


Conclusion: Most if not all plastic parts will absorb moisture >> 0.1% weight gain.

JET PROPULSION LABORATORY
Electronic Parts Engineering Office



Moisture Absorption/Desorption for Intel
56 Ld SSOP Package



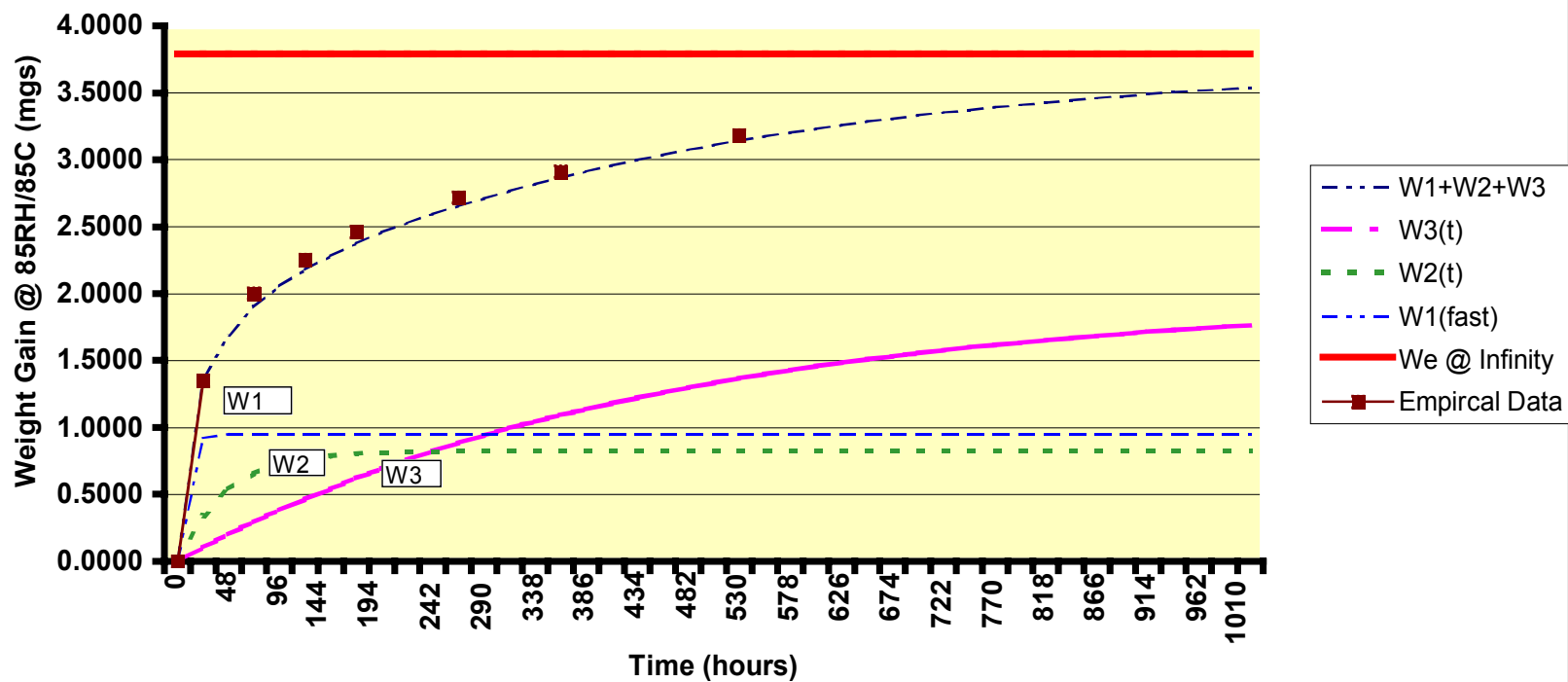
Note: Weight Gain (%) = $(W_t - W_i) / W_i * 100$
Weight Loss (%) = $(W_t - W_i) / (W_f - W_i) * 100$

JET PROPULSION LABORATORY
Electronic Parts Engineering Office



Moisture Multiple Absorption Model for SCR265 (Plastic)

$$W(t) = W_e(1 - e^{-kt})$$



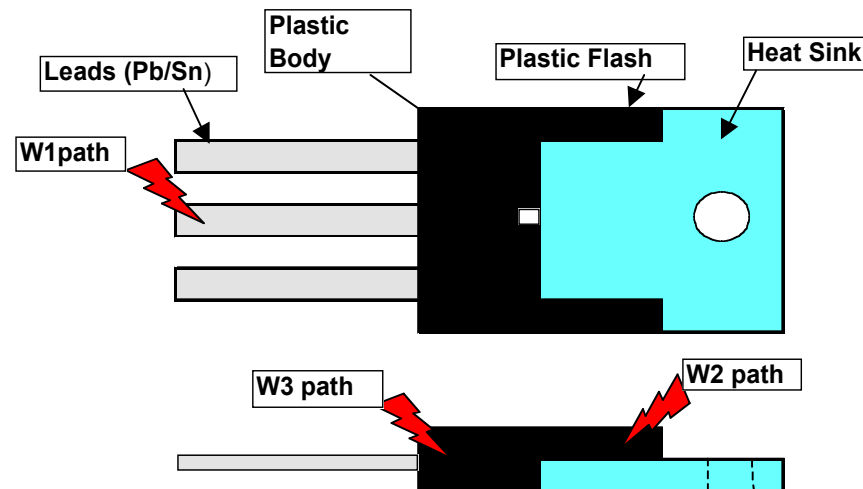
85%RH/85°C Moisture Absorption Mechanisms for SCR265

W1(t): Fast Irreversible Weight Gain ≤ 24 hours ($\sim 1\text{mg}$)

W2(t): Intermediate Reversible Weight Gain, 80 to 140 hours

W3(t): Slowest Reversible Weight Gain, Reaches W_e @ $t=\infty$

SCR265 Package



JET PROPULSION LABORATORY
Electronic Parts Engineering Office

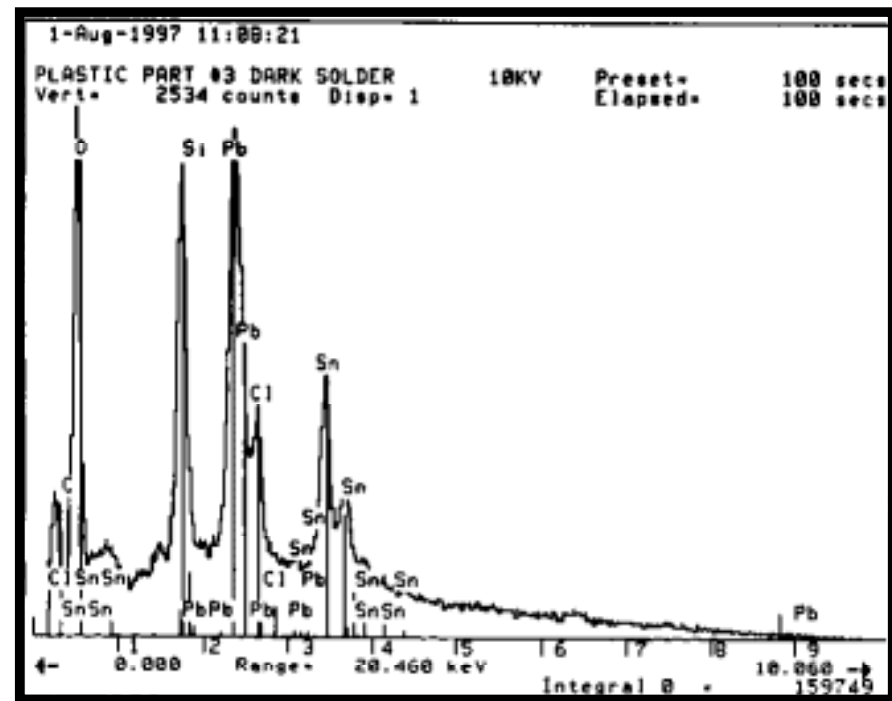
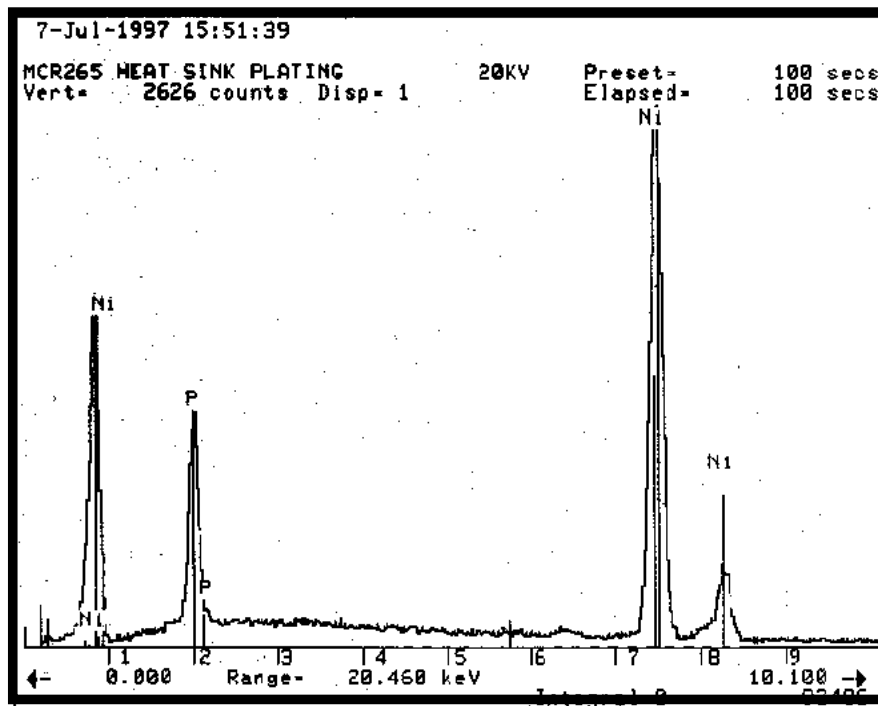


**Nickel Plated Heatsink
Shows No Oxidation**

Post 85%RH/85°C for SCR265

Leads Show Extreme Oxidation

➔ **W1(t)**

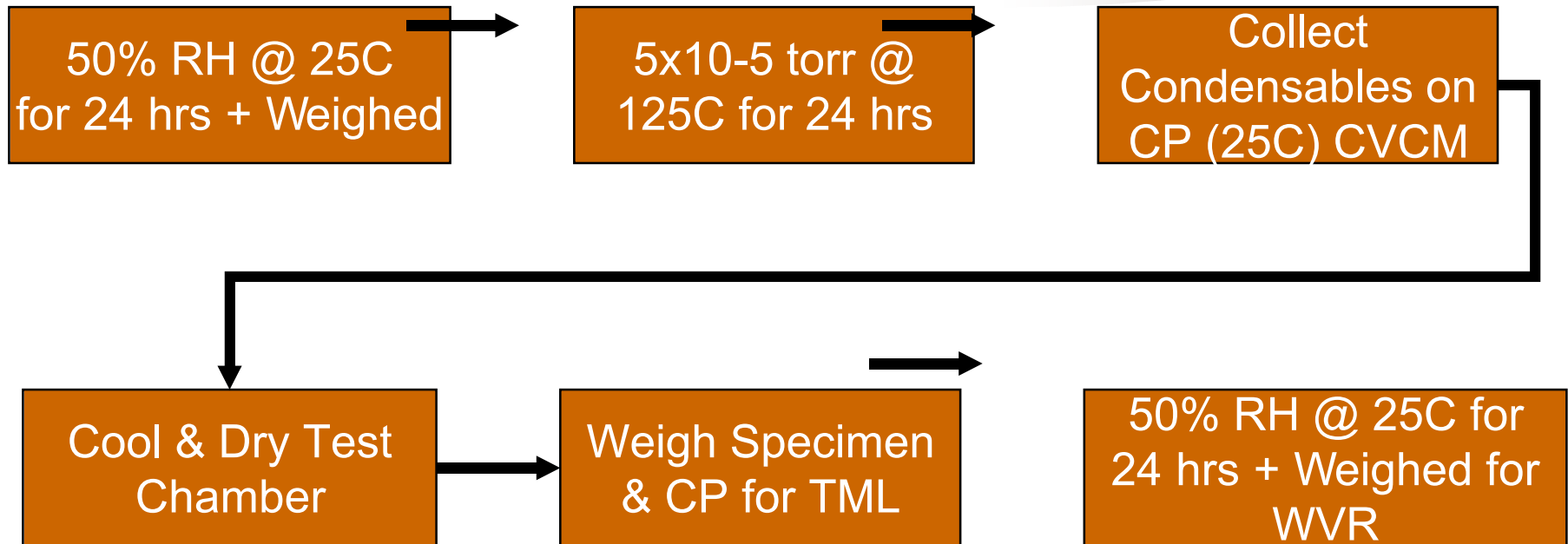


Conclusion: Weight gain is solely attributed to oxidation of leads. The internal chip has miniscule Al area available for oxidation because of Cu intermetallic bonding to the Al.

JET PROPULSION LABORATORY
Electronic Parts Engineering Office



Outgassing Test Flow



Ref: ASTM E595-93

JET PROPULSION LABORATORY
Electronic Parts Engineering Office



Outgassing Results of Plastic Packages

Material	MCR			7612382FBA, E24, DA28F016SV, K8055, U6240332			AM28F020-150PC, 9618FBB			CSI, CAT28F020F, 1-15 09550B		
Part	Motorola SCR			Intel 16 M Flash Memory			AMD 2M Flash Memory			Catalyst 2M Flash Memory		
Sample No.	5	6		7	8	a	9	10		11	24	
WT. Loss %	0.45	0.46	0.45	0.23	0.22	0.22	0.41	0.45	0.43	0.40	0.41	0.40
Water Vapor Recovered, WVR,	0.28	0.25	0.26	0.14	0.11	0.12	0.19	0.17	0.18	0.21	0.18	0.19
%ML (WT, LOSS-WVR) %	0.17	0.21	0.19	0.09	0.11	0.10	0.22	0.28	0.25	0.19	0.23	0.21
CVCM %	0.04	0.08	0.06	0.02	0.01	0.01	0.03	0.05	0.04	0.04	0.04	0.04
DEPOSIT on CP	Opaque			Negligible			Opaque			Opaque		
FTIR Results	Amine cured epoxy			Anhydride cured epoxy			Amine cured epoxy			Amine cured epoxy		

Conclusion: All materials passed. These tests are suited for lot-to-lot comparisons, tracking manufacturing continuity/changes, and measuring absorbed moisture at a known environment.

JET PROPULSION LABORATORY Electronic Parts Engineering Office



Radiation of Plastic Parts

Moisture Absorption / Bake for Intel DA28F016SV in Plastic Package

(0.6 μ m ETOX IV Process Technology)

Conditions: Test Temperature = 25°C, Vdd = 5.0V, Vpp = 5.0V

Dose rate = 25r/s

TID Response of Intel 16M Flash Memory

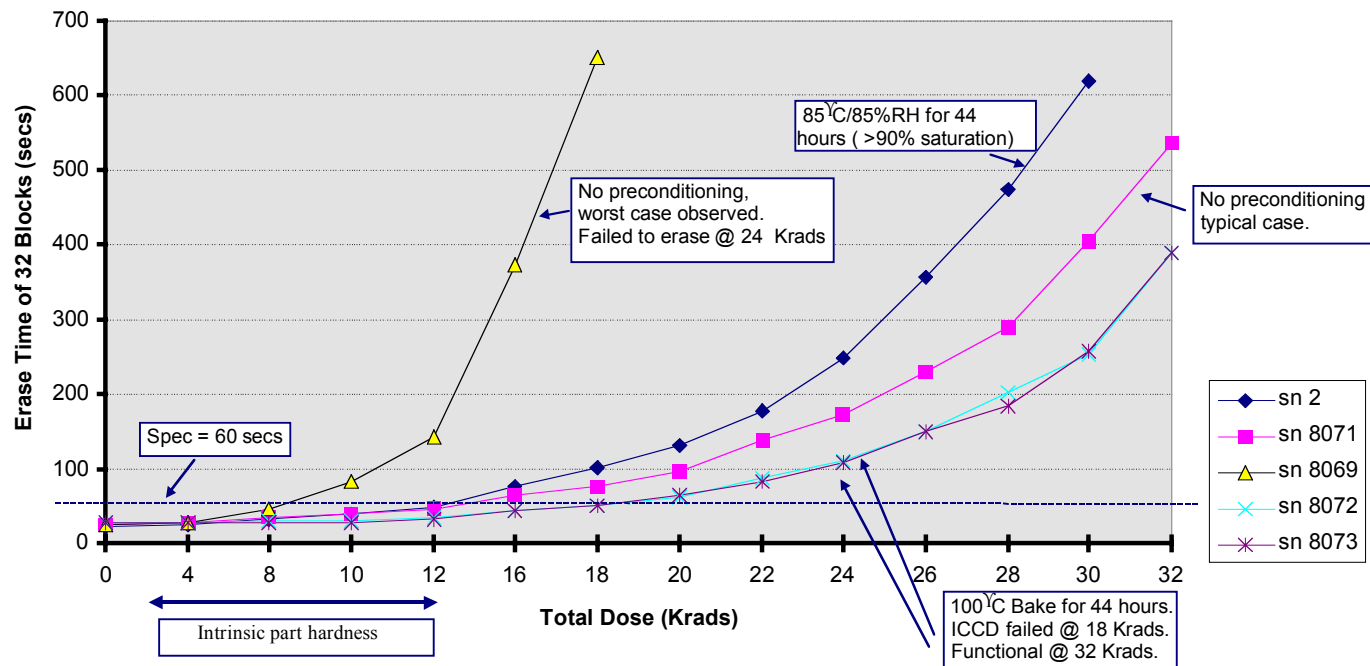


Figure 1
Jet Propulsion Laboratory
Electronic Parts Engineering Office 507



In Summary

- Using plastic parts without understanding their pedigree can lead to mission delay or worst ➡ **Mission Failure**
- A methodology is in place in Office 507 to help JPL users of plastic parts ascertain their risk and acceptance for Space Application
- Work is underway in Office 507 to evaluate **all risk factors** using COTS parts (quality, reliability, radiation, package, and device performance)

JET PROPULSION LABORATORY
Electronic Parts Engineering Office



For Further Information
Contact:

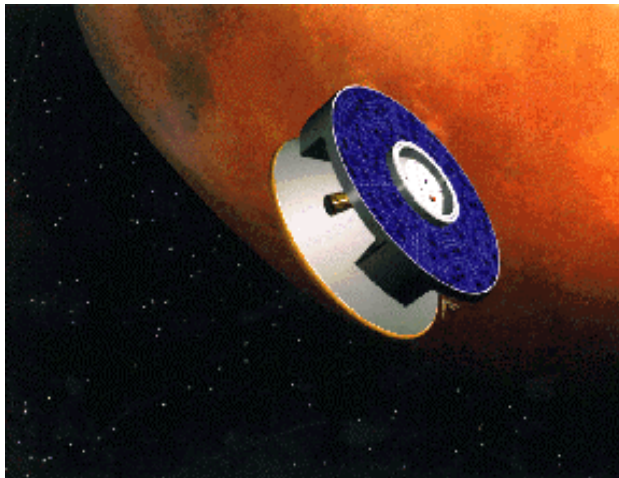
Mike Sandor	Shri Agarwal
or	818-795-4928 x
818-354-0681	203

Technical Contributors:

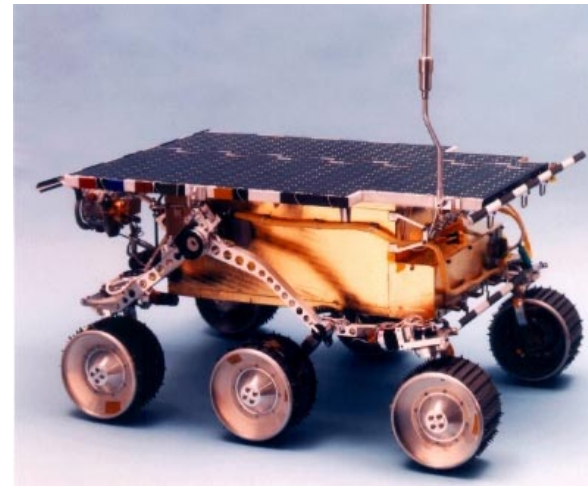
Ed Cuddihy - Jet Propulsion Laboratory
Duc Nguyen - Jet Propulsion Laboratory
Scott McDaniel - DPA Components, Inc.

Plastic Parts Successfully Used For Mars Pathfinder:

16 Mbit DRAM Used in Pathfinder
Flight Computer



FETs ; ASIC & Microcontroller
Used in Modem for Lander and
Rover



Passed 1000 Hours Life Test on Mars !